

## Feedback Controllers and Final Control Elements

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### 1 Steady-state Error

- Steady-state Error Constants
- Steady-State Error for Disturbances
- Open-loop & Feedback Controllers

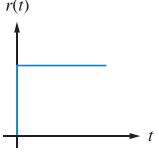
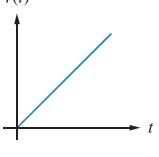
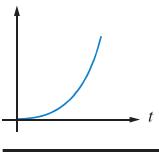
### 2 Response of Controllers



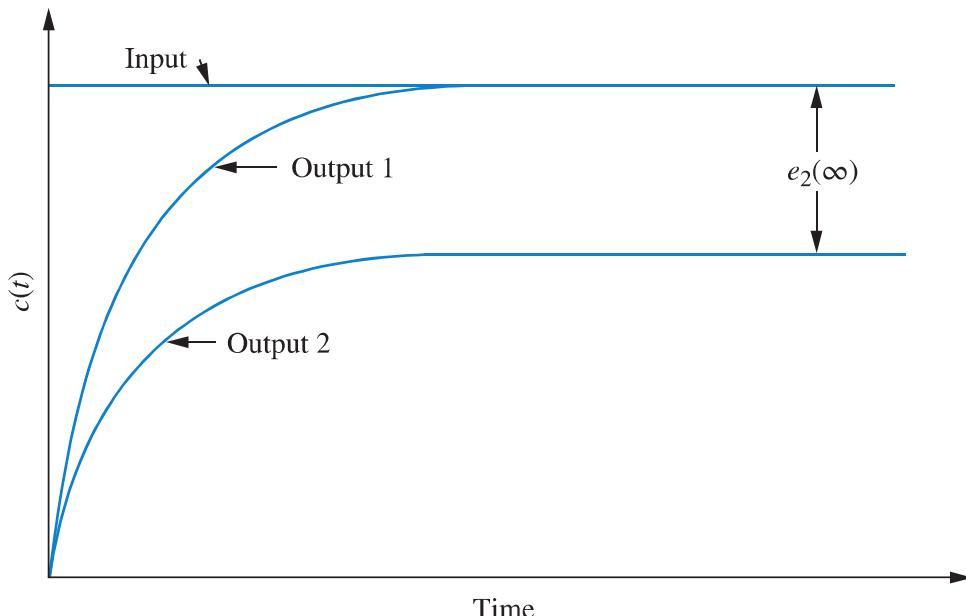
# Steady-state Error

Steady-state error ( $e_{ss}$ ) is the difference between the input and the output for a prescribed test input as  $t \rightarrow \infty$ .

*Test waveforms for steady-state errors of position control systems.*

Waveform	Name	Physical interpretation	Time function	Laplace transform
$r(t)$ 	Step	Constant position	1	$\frac{1}{s}$
$r(t)$ 	Ramp	Constant velocity	$t$	$\frac{1}{s^2}$
$r(t)$ 	Parabola	Constant acceleration	$\frac{1}{2}t^2$	$\frac{1}{s^3}$

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T2044

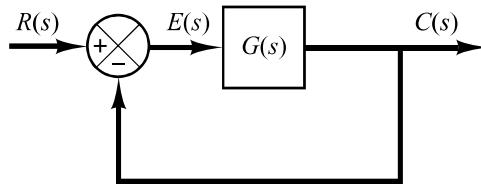
Time

*Steady-state error for step input*



# Steady-state Errors in Feedback Control System

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- $e(t) = r(t) - c(t)$        $\rightarrow E(s) = R(s) - C(s)$
- $\frac{C(s)}{R(s)} = \frac{G(s)}{1+G(s)}$        $\rightarrow \frac{E(s)}{R(s)} = \frac{1}{1+G(s)}$

$$G(s) = \frac{K(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{s^N (T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)}$$

A system is called type 0, type 1, type 2, type p,  
if  $N=0$ ,  $N=1$ ,  $N=2$ ,  $N=p$ , respectively.

- Applying final-value theorem:

$$e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} \frac{sR(s)}{1 + G(s)}$$



## Static Position Error Constant, $K_p$

- Steady-state error for a unit-step input ( $R(s) = 1/s$ ) is:

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s}{1 + G(s)} \frac{1}{s} = \frac{1}{1 + G(0)}$$

$$K_p \equiv \lim_{s \rightarrow 0} G(s) = G(0) \implies e_{ss} = \frac{1}{1 + K_p}$$

- Type 0 system:

$$K_p = \lim_{s \rightarrow 0} \frac{K(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{(T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = K$$

- Type 1 or higher system:

$$K_p = \lim_{s \rightarrow 0} \frac{K(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{s^N (T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = \infty$$

$$e_{ss} = \begin{cases} \frac{1}{1+K} & \text{for type 0 systems,} \\ 0 & \text{for } N \geq 1. \end{cases}$$



## Static Velocity Error Constant, $K_v$

- Steady-state error for a unit-ramp input ( $R(s) = 1/s^2$ ) is:

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s}{1 + G(s)} \frac{1}{s^2} = \lim_{s \rightarrow 0} \frac{1}{sG(s)}$$

$$K_v \equiv \lim_{s \rightarrow 0} sG(s) \implies e_{ss} = \frac{1}{K_v}$$

- Type 0 system:

$$K_v = \lim_{s \rightarrow 0} \frac{sK(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{(T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = 0$$

- Type 1 system:

$$K_v = \lim_{s \rightarrow 0} \frac{sK(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{s(T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = K$$



- For a type 2 or higher system:

$$K_v = \lim_{s \rightarrow 0} \frac{sK(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{s^N(T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = \infty$$

$$e_{ss} = \begin{cases} \infty & \text{for type 0 systems,} \\ \frac{1}{K} & \text{for type 1 systems,} \\ 0 & \text{for type 2 or higher systems.} \end{cases}$$

- Type 0 system is incapable of following a ramp input in the steady state.
- Type 1 system with unity feedback can follow the ramp input with a finite error.
- Type 2 or higher system can follow a ramp input with zero error at steady state.



## Static Acceleration Error Constant, $K_a$

- Steady-state error for a unit-parabolic input ( $R(s) = 1/s^3$ ) is:

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s}{1 + G(s)} \frac{1}{s^3} = \lim_{s \rightarrow 0} \frac{1}{s^2 G(s)}$$

$$K_a \equiv \lim_{s \rightarrow 0} s^2 G(s) \implies e_{ss} = \frac{1}{K_a}$$

- For a type 0 system:

$$K_a = \lim_{s^2 \rightarrow 0} \frac{s K(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{(T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = 0$$

- For a type 1 system:

$$K_a = \lim_{s \rightarrow 0} \frac{s^2 K(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{s(T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = 0$$



- For a type 2 system:

$$K_a = \lim_{s \rightarrow 0} \frac{s^2 K(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{s^2(T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = K$$

- For a type 3 or higher system:

$$K_a = \lim_{s \rightarrow 0} \frac{s^2 K(T_a s + 1)(T_b s + 1) \cdots (T_m s + 1)}{s^N(T_1 s + 1)(T_2 s + 1) \cdots (T_p s + 1)} = \infty$$

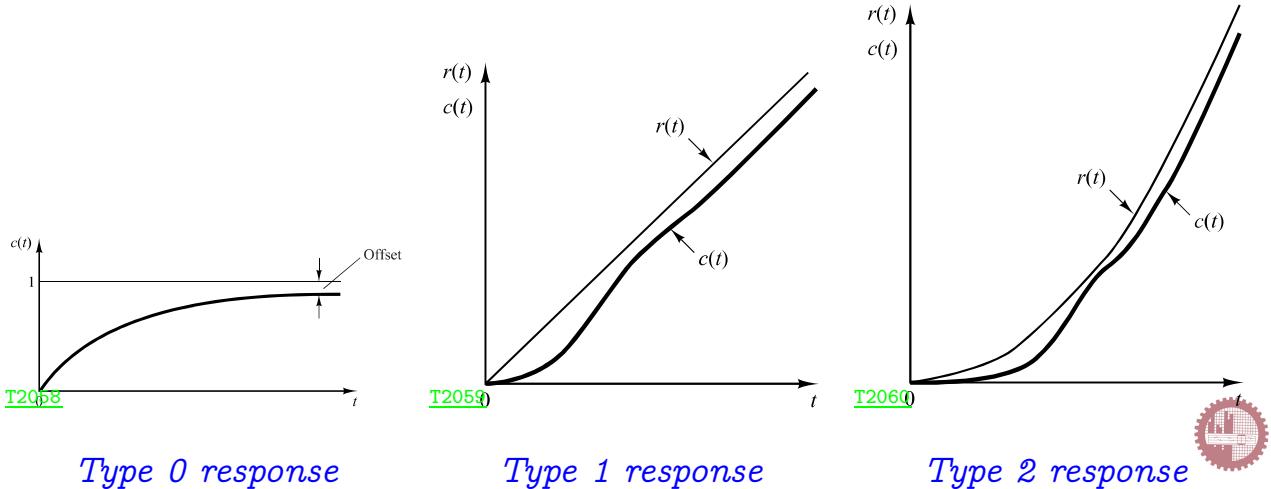
$$e_{ss} = \begin{cases} \infty & \text{for type 0 \& type 1 systems,} \\ \frac{1}{K} & \text{for type 2 systems,} \\ 0 & \text{for type 3 or higher systems.} \end{cases}$$

- Both type 0 and type 1 systems are incapable of following a parabolic input in the steady state.
- Type 2 system with unity feedback can follow a parabolic input with a finite error signal.
- Type 3 or higher system with unity feedback follows a parabolic input with zero error at steady state.

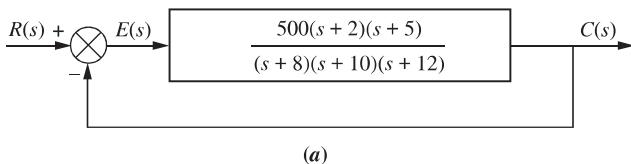


Steady-state error ( $e_{ss}$ )T2049

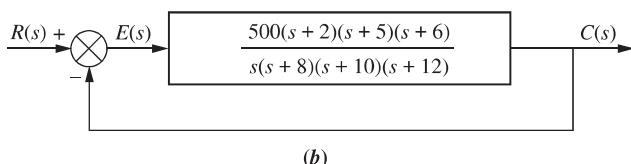
	Step Input $r(t) = 1$	Ramp Input $r(t) = t$	Acceleration Input $r(t) = \frac{1}{2}t^2$
Type 0 system	$\frac{1}{1 + K}$	$\infty$	$\infty$
Type 1 system	0	$\frac{1}{K}$	$\infty$
Type 2 system	0	0	$\frac{1}{K}$



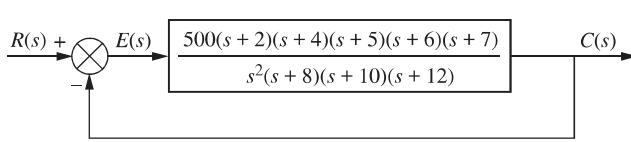
Example: ▷ Steady-State Error via Static Error Constants for standard step, ramp and parabolic input.



- (a):
- ▶ step input,  $e_{ss} = 0.161$
  - ▶ ramp input,  $e_{ss} = \infty$
  - ▶ parabolic input,  $e_{ss} = \infty$



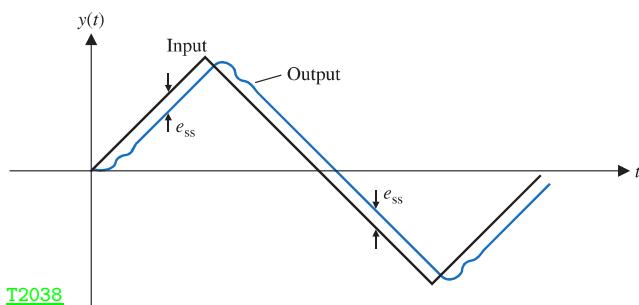
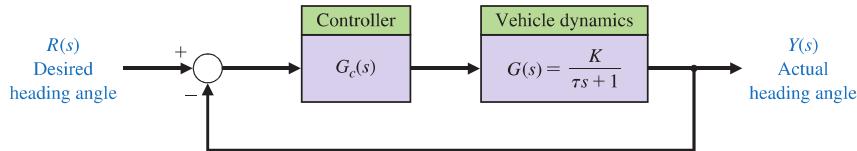
- (b):
- ▶ step input,  $e_{ss} = 0$
  - ▶ ramp input,  $e_{ss} = 0.032$
  - ▶ parabolic input,  $e_{ss} = \infty$



- (c):
- ▶ step input,  $e_{ss} = 0$
  - ▶ ramp input,  $e_{ss} = 0$
  - ▶ parabolic input,  
 $e_{ss} = 1.14 \times 10^{-3}$

T2047

Example: ▷ Mobile robot steering control,  $G_c = K_1 + \frac{K_2}{s}$ .



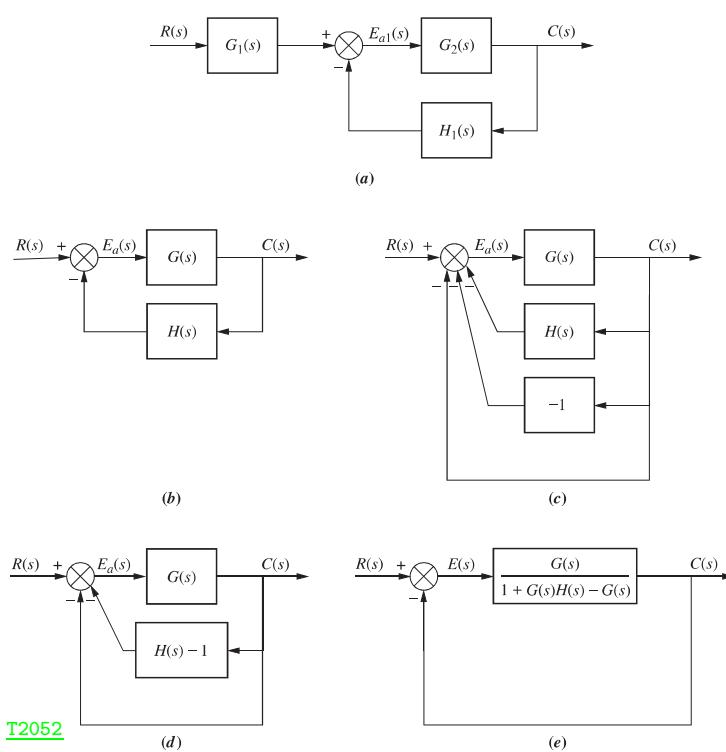
- $G(s) = \frac{K(K_1 s + K_2)}{s(\tau s + 1)}$

- type 1 system.

- For step input:  $e_{ss} = \frac{1}{1 + K_p} = 0$ ;  $K_p = \lim_{s \rightarrow 0} G(s) = \infty$
- For ramp input:  $e_{ss} = \frac{1}{K_v}$ ;  $K_v = \lim_{s \rightarrow 0} sG(s) = K_2 K$
- For a triangular wave input: plot above.



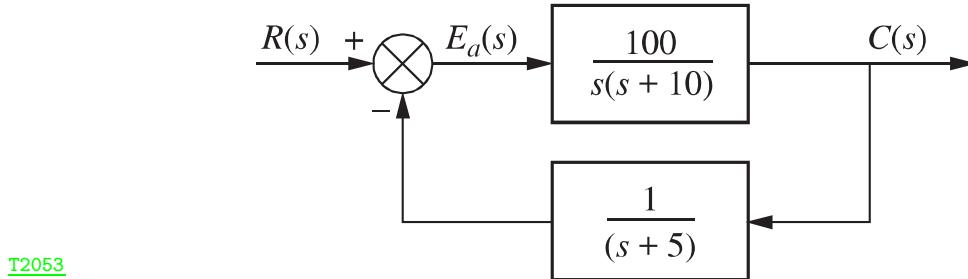
## Steady-State Error for Non-unity Feedback Systems



- Fig.(b):  
 $G(s) = G_1(s)G_2(s)$ , and  
 $H(s) = H_1(s)/G_1(s)$ .
- Fig.(c): form a unity feedback system by adding and subtracting unity feedback paths.
- Fig.(d): combine  $H(s)$  with the negative unity feedback.
- Fig.(e): combine the feedback system consisting of  $G(s)$  and  $[H(s) - 1]$ .



Example: ▷ Steady-State Error for Non-unity Feedback Systems, step input.

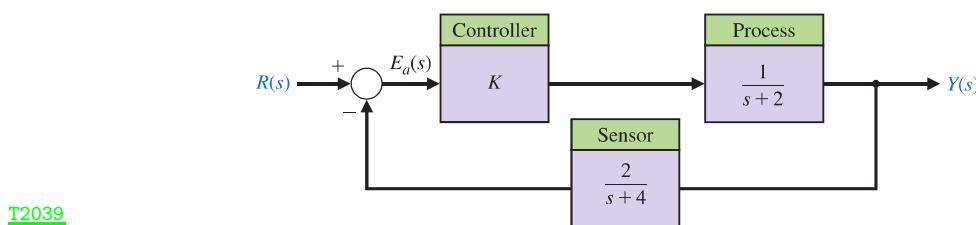


$$G_e = \frac{G(s)}{1 + G(s)H(s) - G(s)} = \frac{100(s+5)}{s^3 + 15s^2 - 50s - 400}$$

- No pure integrator, type 0.
- $K_p = \lim_{s \rightarrow 0} G_e(s) = -5/4$ .
- $e_{ss} = \frac{1}{1+K_p} = -4$ .
- Negative value for steady-state error implies that the output step is larger than the input step.



Example: ▷ Nonunity feedback control system: Estimate  $K$  for  $e_{ss} = 0$ .

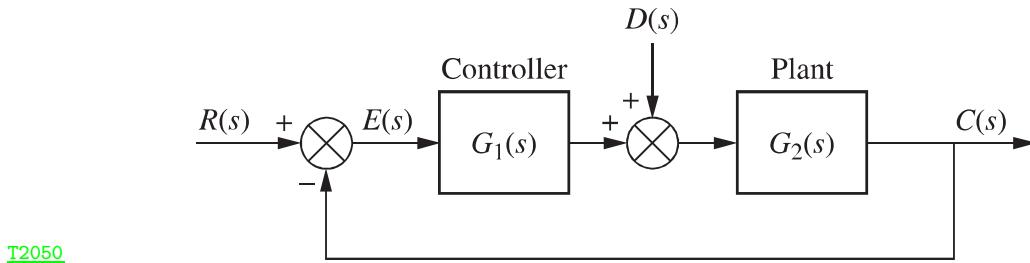


$$G_e = \frac{G(s)}{1 + G(s)H(s) - G(s)} = \frac{K(s+4)}{s^2 + (6-K)s + (8-2K)}$$

- No pure integrator, type 0.
- $K_p = \lim_{s \rightarrow 0} G_e(s) = \frac{4K}{8-2K}$ .
- $e_{ss} = \frac{1}{1+K_p} = \frac{8-2K}{8+2K}$ .
- For  $e_{ss} = 0 \rightarrow K = 4$ .



## Steady-State Error for Disturbances



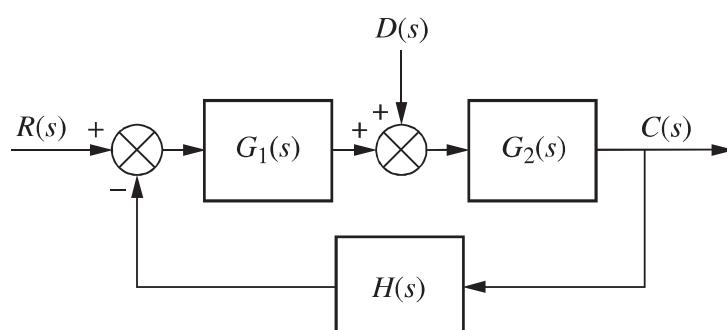
- $C(s) = E(s)G_1(s)G_2(s) + D(s)G_2(s)$ :  $C(s) = R(s) - E(s)$
- $\Rightarrow E(s) = \frac{1}{1+G_1(s)G_2(s)}R(s) - \frac{G_2(s)}{1+G_1(s)G_2(s)}D(s)$

$$e_{ss} = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} \frac{s}{1+G_1(s)G_2(s)} R(s) - \lim_{s \rightarrow 0} \frac{sG_2(s)}{1+G_1(s)G_2(s)} D(s)$$

- $e_{ss,R} = \lim_{s \rightarrow 0} \frac{s}{1+G_1(s)G_2(s)} R(s)$ , steady-state error due to  $R(s)$ ,
- $e_{ss,D} = \lim_{s \rightarrow 0} \frac{sG_2(s)}{1+G_1(s)G_2(s)} D(s)$ , steady state error due to disturbance.



## Non-unity feedback control system with disturbance



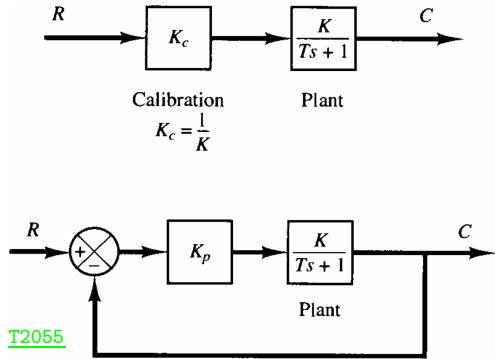
- $e_{ss} = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} s \left[ \left\{ 1 - \frac{G_1(s)G_2(s)}{1+G_1(s)G_2(s)H(s)} \right\} R(s) - \left\{ \frac{G_2(s)}{1+G_1(s)G_2(s)H(s)} \right\} D(s) \right]$
- For a step input and step disturbance:  

$$e_{ss} = \left\{ 1 - \frac{\lim_{s \rightarrow 0} G_1(s)G_2(s)}{\lim_{s \rightarrow 0} [1+G_1(s)G_2(s)H(s)]} \right\} - \left\{ \frac{\lim_{s \rightarrow 0} G_2(s)}{\lim_{s \rightarrow 0} [1+G_1(s)G_2(s)H(s)]} \right\}$$
- For zero error:
  - $\frac{\lim_{s \rightarrow 0} G_1(s)G_2(s)}{\lim_{s \rightarrow 0} [1+G_1(s)G_2(s)H(s)]} = 1$ , and
  - $\frac{\lim_{s \rightarrow 0} G_2(s)}{\lim_{s \rightarrow 0} [1+G_1(s)G_2(s)H(s)]} = 0$ .



# Comparison: Open-loop & Feedback Controllers

$$e(t) = r(t) - c(t) \implies E(s) = R(s) - C(s)$$



### Open-loop system:

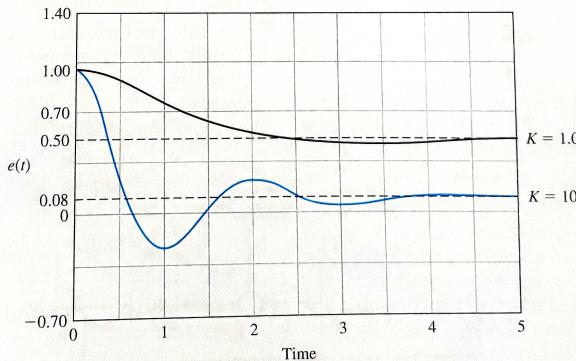
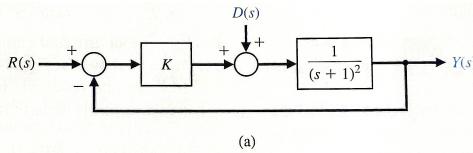
- By calibration,  $K_c = 1/K \rightarrow KK_c = 1: \implies e_{ss} = 1 - G(0) = 0$
- Calibration may drift with time.
  - ▶ if  $K \implies (K + \Delta K)$ ; &  $K = 10, \Delta K = 1:$
  - ▶  $e_{ss} = 1 - K_c(K + \Delta K) = 1 - KK_c(1 + \frac{\Delta K}{K}) = 1 - 1.1 = -0.1 = 10\%$
  - ⇒ Significant change.

### Feedback system:

- Often  $KK_p = 100 \rightarrow e_{ss} = \frac{1}{1+100} = 0.0099 = 9.9\%.$ 
  - ▶ if  $K \implies (K + \Delta K)$ ; &  $K = 10, \Delta K = 1:$
  - ▶  $e_{ss} = \frac{1}{1+K_p(K+\Delta K)} = \frac{1}{1+\frac{K_p}{K}(1+\frac{\Delta K}{K})} = \frac{1}{1+100(1.1)} = 9.0\%$
  - ⇒ Not significant change.



Example: ▷ Analyse Open-loop & Feedback Control Systems,  $R(s) = 0$ .



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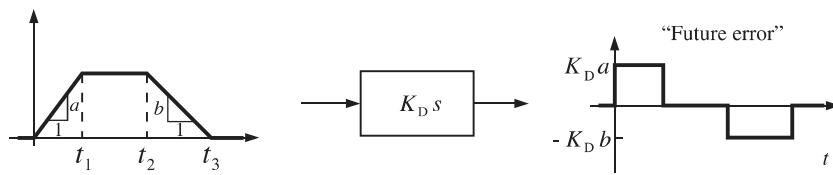
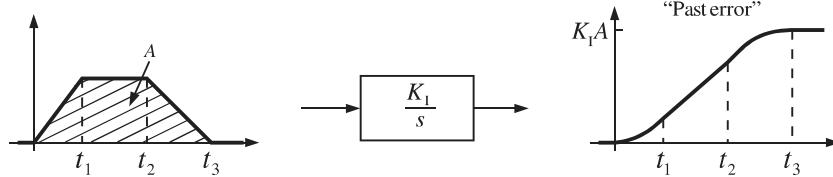
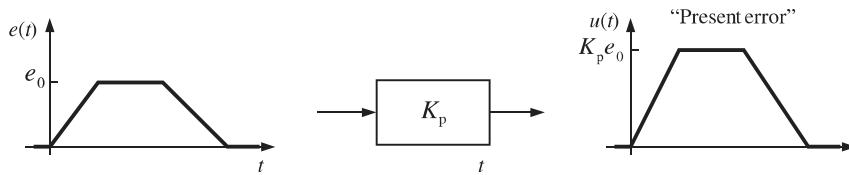
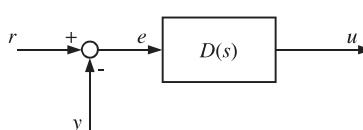
	Open Loop*		Closed Loop		
	$K = 1$	$K = 1$	$K = 8$	$K = 10$	
Rise time (s) (10% to 90% of final value)	3.35	1.52	0.45	0.38	
Percent overshoot (%)	0	4.31	33	40	
Final value of $y(t)$ due to a disturbance, $D(s) = 1/s$	1.0	0.50	0.11	0.09	
Percent steady-state error for unit step input	0	50%	11%	9%	
Percent change in steady-state error due to 10% decrease in $K$	10%	5.3%	1.2%	0.9%	

T2057



### Response of Controllers

## Input-Output Behaviour of Basic Control Actions

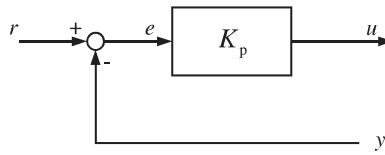


T2035



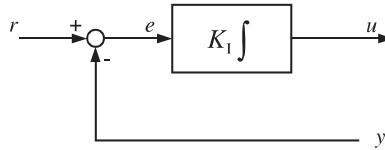
# Basic Feed-back Control Actions

Proportional control (P)



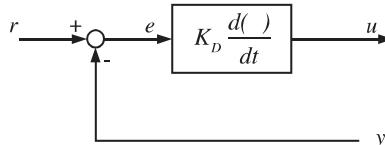
$$\begin{aligned} u(t) &= K_p e(t) \\ D(s) &= \frac{u(s)}{e(s)} = K_p \end{aligned}$$

Integral control (I)



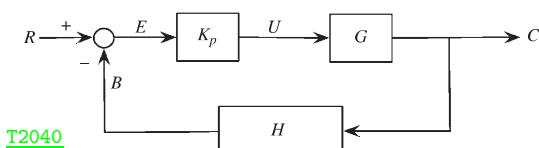
$$\begin{aligned} u(t) &= K_I \int_0^t e(\tau) d\tau \\ D(s) &= \frac{u(s)}{e(s)} = \frac{K_I}{s} \end{aligned}$$

Derivative control (D)

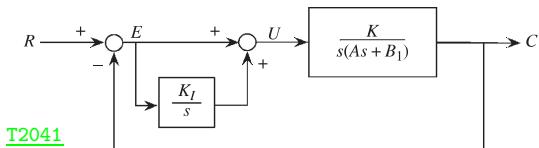


$$\begin{aligned} u(t) &= K_D \frac{d(e(t))}{dt} \\ D(s) &= \frac{u(s)}{e(s)} = K_D s \end{aligned}$$

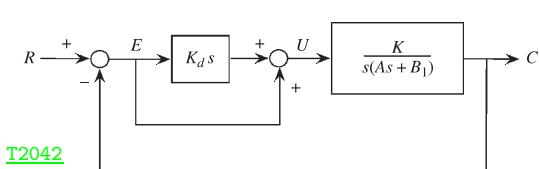
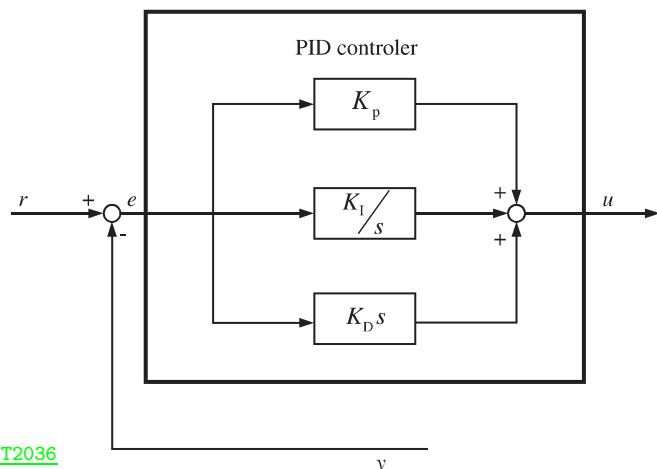
[T2034](#)



*System including P control*



*System including I control*



*System including D control*

$$U(s) = (K_p + K_I \frac{1}{s} + K_D s) E(s)$$

$$D(s) = K_p + K_I \frac{1}{s} + K_D s = K_p \left(1 + \frac{1}{T_i s} + T_D s\right)$$

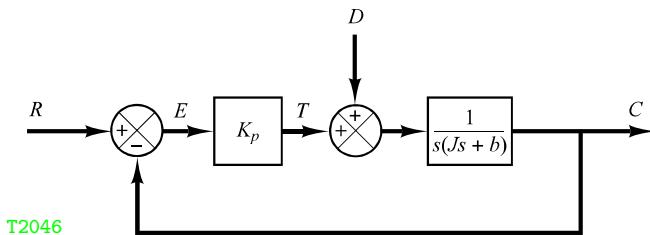
$$K_i = \frac{K_p}{T_i}, \quad K_d = K_p T_D$$



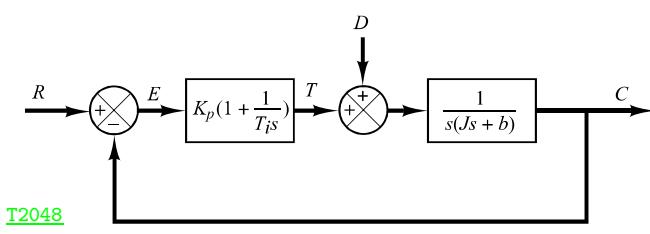
# Response of P, PI Controllers

For a step disturbance,  $D = 1/s$  &  $H(s) = 1$ :

$$e_{ss,D} = \frac{\lim_{s \rightarrow 0} G_2(s)}{\lim_{s \rightarrow 0}[1 + G_1(s)G_2(s)H(s)]} = -\frac{1}{\lim_{s \rightarrow 0} \frac{1}{G_2(s)} + \lim_{s \rightarrow 0} G_1(s)}$$



- For P:  $e_{ss,D} = -1/K_p$
- For PI:  $e_{ss,D} = 0$



Example: ▷ Steady-State Error Due to Step Disturbance

